

SPACE DEBRIS SYMPOSIUM (A6)
Measurements (1)

Author: Dr. Yukihiro Kitazawa
IHI Corporation, Japan

Mr. Haruhisa Matsumoto
JAXA, Japan

Mr. Osamu Okudaira
Japan Aerospace Exploration Agency (JAXA), Japan

Dr. Yugo Kimoto
Japan Aerospace Exploration Agency (JAXA), Japan

Ms. Pauline FAURE
Kyushu Institute of Technology, Japan

Dr. Maki Hattori
University of Tokyo, Japan

Prof. Toshiya Hanada
Kyushu University, Japan

Prof. Dr. Yasuhiro Akahoshi
Kyushu Institute of Technology, Japan

Mr. Atsushi Karaki
IHI Corporation, Japan

Prof. Akira Sakurai
Institute for Q-shu Pioneer of Space, Inc. (iQPS), Japan

Mr. Funakoshi Kunihiro
Institute for Q-shu Pioneer of Space, Inc. (iQPS), Japan

Prof. Tetsuo Yasaka
QPS Institute, Japan

R&D ON IN-SITU SENSORS FOR MMOD MEASUREMENT AT JAXA

Abstract

The history of Japanese R&D into in-situ sensors for micro-meteoroid and orbital debris (MMOD) measurements is neither particularly long nor short. Research into active sensors started for the meteoroid observation experiment on the HITEN (MUSES-A) satellite of ISAS/JAXA launched in 1990, which had MDC (Munich Dust Counter) on-board sensors for micro meteoroid measurement. The main purpose behind the start of passive sensor research was SOCCOR, a late 80's Japan-US mission that planned to capture cometary dust. Although this mission was canceled, the research outcomes were employed in a JAXA micro debris sample return mission using calibrated aerogel involving the Space Shuttle and the International Space Station. There have been many other important activities apart from the above, and the knowledge generated from them has contributed to JAXA's development of a new type of active dust sensor. JAXA and its partners have been developing a simple in-situ active dust sensor of a new type to detect dust particles ranging from a hundred micrometers to several millimeters. The distribution and flux of the debris in the size range are not well understood and is difficult to measure using ground observations. However, it is important that the risk caused by such debris is assessed. In-situ measurement of debris in

this size range is useful for 1) verifying meteoroid and debris environment models, 2) verifying meteoroid and debris environment evolution models, and 3) the real time detection of explosions, collisions and other unexpected orbital events. Multitudes of thin, conductive copper strips are formed at a fine pitch of 100 μm on a film 12.5 μm thick of nonconductive polyimide. An MMOD particle impact is detected when one or more strips are severed by being perforated by such an impact. This sensor is simple to produce and use and requires almost no calibration as it is essentially a digital system. Based on this sensor technology, the Kyushu Institute of Technology (KIT) has designed and developed an educational version of the sensor, which is currently on board the nano-satellite Horyu-II, which was built at KIT and launched on May 18, 2012. Although the sensor has a very small sensing area, sensor data were nonetheless successfully received. JAXA's flight version, to be employed on satellites and/or the ISS, will be ready soon. This paper reports on the RD into in-situ measurement MMOD sensors at JAXA.