

SYMPOSIUM ON SAFETY, QUALITY AND KNOWLEDGE MANAGEMENT IN SPACE
ACTIVITIES (D5)

Space Weather Prediction and Effects on Space Missions (3)

Author: Mr. NOOR DANISH AHRAR MUNDARI
Kyushu Institute of Technology, Japan, g586406d@tobata.isc.kyutech.ac.jp

Dr. Arifur Khan

Japan, markhan@ele.kyutech.ac.jp

Dr. Minoru Iwata

Kyushu Institute of Technology, Japan, iwata@ele.kyutech.ac.jp

Prof. Kazuhiro Toyoda

Kyushu Institute of Technology, Japan, toyoda@ele.kyutech.ac.jp

Prof. Mengu Cho

Kyushu Institute of Technology, Japan, cho@ele.kyutech.ac.jp

UNDERSTANDING THE EFFECT OF ATOMIC OXYGEN EXPOSURE ON SURFACE AND
VOLUME RESISTIVITY CHANGE IN LEO**Abstract**

Spacecraft surface charging can lead to arcing and a loss of electricity generation capability of solar panel or even loss of a satellite. The arcing problem may be further aggravated by the atomic oxygen (AO) exposure in Low Earth orbit, which modifies the surface of materials like Kapton, Teflon, anti reflective coating, cover glass etc used on satellite surface affecting material properties such as resistivity, secondary electron emissivity, photo electron emission which govern the charging behavior. These properties are crucial input parameters for spacecraft charging analysis. To study the AO exposure effect on the charging governing properties, an AO exposure facility based on laser detonation of oxygen was built which produces 8-14km atomic oxygen with higher flux than in orbit. After exposing materials to 10 years equivalent AO flux at the altitude of 700-800km, surface charging properties like resistivity and secondary electron emission are being measured. Measurement of surface and volume resistivity for AO exposed samples is under progress initially being done for Kapton. The measurement is being done using the charge storage decay method at a room temperature, which is considered most appropriate for measuring the resistivity for space application. The experimental analysis of material properties will be presented in conference.