

IAF SPACE EXPLORATION SYMPOSIUM (A3)
Mars Exploration – missions current and future (3A)

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MODELING A MARS LOX/LH₂ ARCHITECTURE WITH CRYO-MANAGEMENT, ISRU, AND FUEL
CELLS**Abstract**

LOX/LH₂ Mars architectures have historically been ruled out in favor of *LOX/CH₄* architectures, which leaves a gap in literature and technology development in support of *LH₂* architectures. This leads to a perceived risk with long-term *LH₂* cryo-management and fuel ISRU, even though discoveries suggest that water is widely accessible on Mars. We model a *LOX/LH₂* Mars architecture with passive and active cryo-technologies, fuel cells to produce water and power from the otherwise wasted boiloff, and fuel ISRU.

As these technologies are switched on and off in combinations, the resulting mass that has to be sent to Earth's orbit, or upmass, is compared with a constant payload is assumed to go to the surface. Alternatively, another analysis shows how much payload could be achieved given a constant upmass. With improved MLI and cryocoolers, a 5% – 10% upmass saving was achieved, or an increase in landed mass from 22 to 28mT. The addition of fuel cells only gave a marginal benefit of 0.7% upmass savings, but they add operational flexibility for power and life support. Further, the added mass due to Mars water ISRU infrastructure pays off due to mass savings just after the second crewed mission. These results suggest a Mars *LOX/LH₂* architecture with improved MLI and cryocoolers will be competitive with a *LOX/LCH₄* architecture.