IAF SPACE POWER SYMPOSIUM (C3) Space Power System for Ambitious Missions (4)

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MODULAR DESIGN OF A SPACE MW-LEVEL POWER SYSTEM USING A MOLTEN SALT REACTOR

Abstract

Currently developing projects with ambitious goals such as lunar orbital stations, moon bases, multimission solar system explorations and Mars human exploration raise the question of dense, controllable, reliable, safe and powerful power supply.

Present work consists in a modular proposition of a MW level power system using a molten salt reactor as primary heat source. We studied different options for the energy conversion system: a potassium or a mercury Rankine cycle combined with a simplistic Heat Pipe Radiator (HPR) or a Liquid-Solid Droplet Radiator (LSDR) as proposal for the heat sink. A simple study of the anti-radiation shield has also been lead.

This paper proposes two configurations for two applicative cases, from a performance point of view, and discusses modulations of the global system for others applications. The criteria of specific mass has been chosen as performance indicator, since it represents a relevant indicator for any space power supply.

First case concerns an unmanned space probe, second is a manned spacecraft. Both cases share the same reactor design and weighs around 1.5 tons. Shield results from radiation limits considerations and weighs about 1 tons. Proposed conversion system with the corresponding heat sink results from an optimization study on various Rankine cycles and heat sinks combinations. A Rankine cycle using potassium or mercury as working fluid, coupled with a Heat Pipes Radiator (HPR) or a Liquid-Solid Droplet Radiator (LSDR) have been analyzed. The optimal option is a combination of a high temperature potassium Rankine cycle combined with a eutectic aluminum-germanium LSDR. This proposed option weighs about 6 tons, reaching 8.5 kg/kWe.

Despite the two detailed proposals, it is relevant to put things into perspective. LSDR and potassium Rankine are in fact lesser-developed technologies. Thus, the idea of modularity: for a swifter implementation, the combination of a mercury cycle with a water heat pipes radiator would be better, although not optimal from the specific mass criteria doubling it to 16.5 kg/kWe. Likewise, the previously proposed shield design reuses long dated ideas, structure and materials. Other designs currently under investigation may lead to a simpler shield, although heavier.