IAF SPACE PROPULSION SYMPOSIUM (C4)
Joint Session between IAA and IAF for Small Satellite Propulsion Systems (8-B4.5A)

Author: Dr. Fabrizio Stesina
Politecnico di Torino, Italy, fabrizio.stesina@polito.it

Prof. Sabrina Corpino
Politecnico di Torino, Italy, sabrina.corpino@polito.it Dr. Daniele Pavarin
CISAS - "G. Colombo" Center of Studies and Activities for Space, University of Padova, Italy, pavarin@dim.unipd.it
Mr. Eduard Bosch Borràs
ESA, The Netherlands, eduard.bosch.borras@esa.int
Mr. Jose Gonzalez del Amo
European Space Agency (ESA), The Netherlands, Jose.gonzalez@esa.int
Dr. Nicolas Bellomo
T4i, Italy, n.bellomo@t4innovation.com
Dr. Fabio Trezzolani
T4i, Italy, f.trezzolani@t4innovation.com
Dr. Marco Manente
T4i, Italy, m.manente@t4innovation.com
Dr. Alessandro Barbato
T4i, Italy, a.barbato@t4innovation.com
Ms. Luisa Iossa
Politecnico di Torino, Italy, luisa.iossa@studenti.polito.it
Mr. Vincenzo Calabretta
Polytechnic of Turin, Italy, vincenzo.calabretta@studenti.polito.it
Mr. Domenico Parrinello
Politecnico di Torino, Italy, domenico.parrinello@polito.it
Mr. Simone Cantarella
Politecnico di Torino, Italy, s290654@studenti.polito.it
Dr. Matteo Duzzi
T4i, Italy, matteo.duzzi@gmail.com

## ENVIRONMENTAL TEST CAMPAIGN OF A 12U CUBESAT TEST PLATFORM EQUIPPED WITH AN AMBIPOLAR PLASMA THRUSTER


#### Abstract

CubeSats are becoming important space platforms for achieving a broad set of mission goals, such as science and Earth observation and technology demonstration. These missions require the improvement of the CubeSat capabilities, such as orbit change and raising, formation flying, and proximity operations. In this sense, miniaturised electric propulsion systems (ePS) are valuable enabling technology. To increase the ePS readiness level, it is essential to assess the mutual interactions between the ePS and the other onboard systems favouring the integration of the ePS at system level. The paper proposes the results obtained


during the test campaign in relevant environment performed by a 12 U CubeSat Test Platform (CTP), developed in the ESA-Prop roject, supported by the ESA Propulsion Laboratory and led by Politecnico di Torino, with REGULUS-50-Xe, an Ambipolar Plasma Thruster designed and developed by T4i. CTP has a 12 U primary structure, where up to 8 Us can accommodate the ePS and the remaining 4 U the CubeSat subsystems and two 160 Wh batteries. CTP gathers unprecedented information on the mutual impact of propulsion system on the CubeSat technology and vice versa. Examples are internal thermal fluxes, power consumption in different modes of operation of the platform and ePS, electromagnetic field generation and magnetic fields. CTP is equipped with RF sensing circuits (tuned in different frequency ranges), Line Impedance Stability Network, temperature sensors, current and voltage sensing circuits, and magnetometers. These measurements are fused in post processing with the acquisitions made by sensors and instruments, such as Faraday cups and Magnetic Field Mapper, installed outside the CTP and inside the test chamber. Moreover, CTP manages the housekeeping data acquisition and handling, commands execution, power distribution, battery recharging, and communication via wired and radiofrequency links with the ground support system (GSS) where the operators control the test operations. The paper describes the integration process of REGULUS-50-Xe inside CTP and the tests in laboratory conditions performed in the University of Padova Plasma Propulsion Facility and it shows the results of the environmental test performed in the Small Plasma Facility at ESTEC, that includes short and long duration firing at different power levels.

