IAF SPACE TRANSPORTATION SOLUTIONS AND INNOVATIONS SYMPOSIUM (D2) Upper Stages, Space Transfer, Entry & Landing Systems (3)

Author: Mr. Zhigang ZHANG CAS Space (Guangzhou Zhongke Aerospace Exploration Technology Co., Ltd.), China, zhangzhigang@caspace.com.cn

Mrs. Zhijing ZHANG

CAS Space (Guangzhou Zhongke Aerospace Exploration Technology Co., Ltd.), China, zhangzhijing@caspace.com.cn Mr. Weiping Wu CAS Space (Guangzhou Zhongke Aerospace Exploration Technology Co., Ltd.), China, wuweiping@caspace.com.cn Mr. Yuhai MA CAS Space (Guangzhou Zhongke Aerospace Exploration Technology Co., Ltd.), China, mayuhai@caspace.com.cn

LANDING GUIDANCE METHOD FOR REUSABLE LAUNCH VEHICLE BASED ON CONVEX OPTIMIZATION

Abstract

Online autonomous fight trajectory planning according to real-time mission requirements is a key problem to be solved for reusable launch vehicle fixed-point landing missions. Complex constraints such as control constraints, path constraints and fixed-point constraints need to be satisfied during the landing process, and the planning algorithm is required to be efficient and optimal. Aiming at the problem, this paper designs a flight path planning and guidance algorithm based on the convex optimization method. A mathematical model of a reusable launch vehicle landing mission is first established, and then the model is transformed with the lossless convexification method. Then a discrete computing model that could be applied in engineering is given. Finally, it is verified by flight tests, and the results show that the method can meet the real-time online computing requirements and achieve high-precision fixed-point landing while minimizing mission fuel consumption. Key Words: reusable launch vehicle; Online trajectory planning; Convex optimization; Flight test verification