

IAF SPACE EXPLORATION SYMPOSIUM (A3)
Small Bodies Missions and Technologies (Part 1) (4A)

Author: Dr. Tra Mi Ho

Deutsches Zentrum für Luft- und Raumfahrt e.V. (DLR), Institute of Space Systems, Germany

Dr. Martin Hilchenbach

Max-Planck Institute for Solar Systems Research,, Germany

Mr. Jan Thimo Grundmann

DLR (German Aerospace Center), Germany

Dr. Stephan Ulamec

Deutsches Zentrum für Luft- und Raumfahrt e.V. (DLR), Germany

Dr. Jens Biele

Deutsches Zentrum für Luft- und Raumfahrt e.V. (DLR), Germany

Dr. Frank Dannemann

German Aerospace Center (DLR), Germany

Prof. Bastian Gundlach

Germany

Mr. Henning Fischer

Max-Planck Institute for Solar Systems Research,, Germany

Dr. Carsten Güttler

Max Planck Institute, Germany

Dr. Matthias Grott

DLR (German Aerospace Center), Germany

Prof. Thorsten Kleine

Max-Planck Institute for Solar Systems Research,, Germany

Dr. Norbert Krupp

Max-Planck Institute for Solar Systems Research,, Germany

Mr. Michael Lange

DLR (German Aerospace Center), Germany

Dr. Andreas Nathues

Max-Planck Institute for Solar Systems Research,, Germany

Mr. Dominik Quantius

Deutsches Zentrum für Luft- und Raumfahrt e.V. (DLR), Germany

Dr. Christian J. Renggli

Max-Planck Institute for Solar Systems Research,, Germany

Ms. Nicole Schmitz

Deutsches Zentrum für Luft- und Raumfahrt e.V. (DLR), Germany

Dr. Martin Sippel

Deutsches Zentrum für Luft- und Raumfahrt e.V. (DLR), Germany

Dr. Oliver Stenzel

Max-Planck Institute for Solar Systems Research,, Germany

Dr. Jean-Baptiste Vincent

Deutsches Zentrum für Luft- und Raumfahrt e.V. (DLR), Germany

Mr. Torben Wippermann

Deutsches Zentrum für Luft- und Raumfahrt e.V. (DLR), Germany
Dr. Stephan Theil
Deutsches Zentrum für Luft- und Raumfahrt e.V. (DLR), Germany
Dr. Hans-Ulrich Auster
Technische Universität Braunschweig, Germany
Prof.Dr. Dirk Plettmeier
Technische Universität Dresden (DTU), Germany
Dr. Özgür Karatekin
Royal Observatory of Belgium, Belgium
Mr. Niklas Wendel
German Aerospace Center (DLR), Germany
Mr. Pawel Goldyn
DLR (German Aerospace Center), Germany
Dr. Naomi Murdoch
ISAE-Supaero University of Toulouse, France
Prof. Ferdinand Plaschke
Technische Universität Braunschweig, Germany

SCIENTIFICALLY STRENGTHENING AN ASTEROID MISSION WITH SMALL PROBES ON RAMSES AS USE CASE

Abstract

The close flyby of asteroid (99942) Apophis on 13 April 2029, when it will pass Earth within 38.000 km, presents a unique scientific and strategic opportunity for asteroid research and planetary defense in particular. This event has prompted NASA to revise its OSIRIS-APEX (OSIRIS-REx Extended Mission to Asteroid Apophis) mission to rendezvous Apophis after its closest approach, while ESA is currently studying the RAMSES mission for a rendezvous before Earth's closest approach. One of RAMSES main scientific objectives is to observe the tidal and magnetospheric effects on the Near-Earth Asteroid (NEA) during this close flyby, including possible surface changes. An in-situ science package/lander with a suite of instruments would provide important ground truth and thus improve our understanding of Apophis, especially in determining, for example, its internal structure, magnetic properties, surface features in the micrometer range and volatiles. In addition, returning asteroidal samples to Earth by taking advantage of the short duration of the sample return leg, which requires only a tiny extra boost, is an extremely attractive scenario. We present a lander concept, MASCOT3, based on the heritage of MASCOT flown on the Hayabusa2 mission and MASCOT2, studied for the AIM mission concept as well as a sampler concept, the APOphiS Surface saMpller (APOSSUM), which has undergone a concurrent engineering (CE) feasibility study. The RAMSES mission schedule foresees the rendezvous with the asteroid Apophis about 2 months before its closest approach with Earth in 2029. During this time, early March, the lander or sampling probe will be detached and self-delivered to the asteroid. After a controlled landing, surface operations such as in-situ observations or regolith sampling can be performed. The sampler, after completing its operation, will ascend from the surface of the asteroid and injected on a trajectory to re-enter Earth with a delta-v requirement of only a few tens of meters per second. This is orders of magnitude less than the speed requirement of sample-return missions to other targets, thanks to the very close Earth flyby of Apophis on 13 April 2029. The reentry capsule will return to Earth as the asteroid passes at a safe distance. The reentry capsule's entry velocity is about 12.6 km/s, compared to the asteroid's 7.4 km/s due to Earth's gravitational field. By entering the atmosphere in phase with the Earth's rotation, the entry velocity can be slightly reduced.